AIRWORTHINESS DIRECTIVE COMPLIANCE RECORD

SOCATA 1866 **TB-20** Aircraft, engine, Propeller, or Appliance/Component Make Ser. No. One - Time RECURRING Recurring AD Next Authorized Date and compliance Number & Subject Method of Compliance hours at Signature and due Rev. Date compliance Number AD'S Date/Hours X 98-18-13 Main landing gear bearing @ 4000 DUE @ 4000 hr ACTT @4000 10-24-98 LEFT side Hr ACTT Hr ACTT 05-12-06 X Mag impulse coupling 8-14-2012 PCW by inspection 7-19-05 1854.0 SEE G&N AD LIST 2351.0 01-23-04 Rudder hinge 04-08-2014 PCW by insp lower hinge 1-4-02 2247.0 SEE #1 AF LOG 4-2015 03-04-03 Alleron gimbal joint 04-08-2014 PCW by insp 4-7-03 2247.0 SEE #1 AF LOG 2347.0 X 11-26-04 Fuel injection lines PCW by insp 04-08-2014 1-25-12 2247.0 SEE #2 ENG-LOG 2347.0 X 2005-12-06 INSP IMPULSE 8-19-14 MAGNETO 23461 coupling FOR WEAR 2846 03-04-03 8-19-14 INSP GIPTOAL JOINT 23461 2446 FOR WEAR 11-2670 8-19-14 INSP FUEL LINES 2446 2346. 03-04-03 INSP GIMBAL JOINT AILERON Gimbal 2546.1 WEAR FUEL INJECTION 11-26-04 INSO FUEL LINES 2546. + clamps



RECORD of ACCESSORIES MAJOR PARTS SHIPPED with ENGINE

652 Oliver Street

Williamsport, PA 17701 U.S.A.

Model No: 10-540-C4D5D

Serial No: L-15176-48E

Enpl: HENPL-RT8542

Order No AR339795

TC No: 1E4

Date: 10/23/08

Part Name	Part Number	Manufacturer	Serial	Setting
CARBURETOR				
INJECTOR	73937	PAM	70D41809	2524054-11
LT ALTERNATOR RT ALTERNATOR	32C19553	KELLY	1081714	
MAGNETO LEFT MAGNETO RIGHT MAGNETO DUAL LASER IGN CONT	66E21576	ТСМ	G08EA039	
STARTER	31B22474	KELLY	1052126	
FUEL PUMP	LW-15473	LYC	3608	
IGN HARNESS L IGN HARNESS R LASER IGN HARNESS	67H21475	TCM		
SPARK PLUG	1182-F7	Champion		
LT TURBOCHARGER RT TURBOCHARGER				
BYPASS VALVE				
DENSITY CONTR.				
PRESS. CONTR.				
AB PRESS REL.				
#1 INTERCOOLER #2 INTERCOOLER		æ.		
CRANKCASE MATCH N	NO. K0085	CRANKSHAFT SE	RIAL NO. V13474	

All accessories listed are 0 (zero) time since New or 0 (zero) time since Overhaul. All accessories are new unless part number is succeeded by -85 or -70.

Released: Inspector

Date: 10/23/2008

C of A Issued Date:



NOTE: Form to be used on all New, Overhauled, Rebuilt Engine Models.

Form ET-001 (REV 03/05)



AIRWORTHINESS DIRECTIVE COMPLIANCE LIST

10/23/08

652 Oliver Street

Williamsport, PA 17701 U.S.A.

Engine Model No: IO-540-C4D5D

Engine SN: L-15176-48E

Page 1 of 2

	1	Engine Model No: 10-540-C4D5D	Engine SN: L-15176-48E		rage		01012
AD NO. / SB	RV RV	Description / Method of Compliance	Next Due	Once	Rec	Code	Sign
2002-12-07 543	В	OIL FILTER CONVERTER PLATE GASKET REPLACEMENT NEW P/N CONVERTER PLATE INSTALLED	N/A	х		1	37
2004-10-14 475	С	CRANKSHAFT GEAR MODIFICATION AND ASSY PROCEDURES NEW P/N INSTALLED	N/A	х		1	W
2005-12-06 425	С	Inspection of impulse couplings and stop pins New snap ring configured magneto installed	as per AD		Х	1	Q
2005-19-11 566		Crankshaft replacement Replace Crankshaft	N/A	X		1	
2006-20-09 569	Α	CRANKSHAFT REPLACEMENT REPLACE CRANKSHAFT	N/A	X		1	37
2008-08-14 581		Inspection of Precision Airmotive Hex Plug in regulator cover Inspection per latest revision PRS-107	50 HOURs		Х	1	(I)
2008-14-07 342	E	FUEL LINE AND SUPPORT CLAMP INSPECTION & INSTALL. NEW LINES INST. WITH NEW CLAMP	100 HOURS		Х	1	(3)
63-14-03 295		OIL PUMP DRIVE SHAFT NEW CONF. PART INSTALLED	N/A	X		1	3
66-20-04 307		OIL FILTER ADAPTER GASKET NEW CONF. GASKET INSTALLED	N/A	x		1	(I)
73-23-01 367	F	INSP FOR CRACKS IN PISTON PINS INST NEW PARTS	N/A	х		1	Q'I
75-09-15 382		BENDIX FUEL INJ RS-43 INSP OF MOD OF FLOW DIVIDERS NEW P/N GASKET INSTALLED	N/A	Х		1	0
79-04-05 433	Α	BENDIX FUEL INJ. BULLETIN RS-57 NEW P/N INSTALLED	N/A	x		1	3
87-10-06 477	R1 A	INSPECTION AND REWORK OF LW-18790 ROCKER ARM ASSY NEW CONFIGURATION P/N INST.	N/A	x		1	(H)
92-12-05 501	В	LW-14077 PISTON PIN NEW PARTS INSTALLED	N/A	х		1	(I)
95-07-01 N/A		CONNECTING ROD BOLT FAILURE LYCOMING PART INSTALLED	N/A	X		1	
96-23-03 525	А	HIGH PRESSURE FUEL PUMP INSTALLED LW-15473 NEWLY MANUFACTURED PUMPS INST.	N/A	x		1	
97-15-11 527	С	RECALL OF PISTON PIN P/N LW-14077 NEW CONFIGURATION P/N INST.	N/A	X		1	©
2003-14-03 529	В	ROTARY FUEL PUMP TORQUE. NEW PUMP WITH */M" SUFFIX INSTALLED	N/A	,	Х	2	
١							

Codes:

- 1 AD Applicable to Engine
- 2 AD Not Applicable to Engine
- 3 Field Compliance where applicable





AIRWORTHINESS DIRECTIVE COMPLIANCE LIST

10/23/08

652 Oliver Street

Williamsport, PA 17701 U.S.A.

Engine Model No: IO-540-C4D5D

Engine SN: L-15176-48E

Page 2 of 2

AD NO./	RV	Description /		Once	Rec	Code	Sign
SB	RV	Method of Compliance	Next Due				
2004-05-24		CRANKSHAFT GEAR RETAINING BOLT REPLACEMENT.					
554		NEW BOLT P/N INSTALLED.	N/A	X		2	
64-16-05		AC FUEL PUMP OIL SEAL			- C-1875.) in		
298	Inactive	NEW CONF. FUEL PUMP INSTALLED	N/A	X		2	
67-22-06	4	REPLACEMENT OF BENDIX FUEL DIAPHRAGM ASSY				(A)	
305	В	NEW OR NEWLY O/H INSTALLED	N/A	×		2	
78-23-08		FUEL LINE BETWEEN FUEL PUMP AND FUEL INJECTOR	COV.027V				
421		NEW FLEXIBLE HOSE USED	N/A	×		2	
78-23-10		BENDIX FUEL INJ. BULLETIN RS-42					
428		NEW P/N INSTALLED	N/A	X		2	
81-03-05		BENDIX FUEL INJ BULL. #RS-62 REV 2					
444	Α	NEW P/N INSTALLED	N/A	X		2	
83-22-04		BENDIX BULLETIN #RS-88					
467		BENDIX INJ NOT INSTALLED	N/A	X		2	
94-01-03	R2	DEFECTIVE IGN COILS AND ROTATING MAGNETS					
N/A		NEW CONFIGURATION PART INST.	N/A	X		2	
94-06-09		INCOMPLETE INTERNAL GROUNDING ON MAG CAPACITORS					
517		NEW CONF. CAPACITORS INSTALLED	N/A	×		2	
95-26-02		ENGINES OPERATED WITH LOW OCTANE FUEL					
398		ENGINE IS NEW OR HAS BEEN REBUILT AT MANUFACTURER	N/A	X		2	
96-09-10		REPLACEMENT OF OIL PUMP IMPELLERS					
524		STEEL IMPELLERS INSTALLED	N/A	X		2	
98-17-11		CRANKSHAFT REPAIRED BY NELSON BALANCING SERVICE					
N/A		MAG/PARTICLE INSP OF C/SHAFT	N/A	X		2	
99-04-04		MAGNETO IMPLUSE COUPLING					
537		NEW OR NEWLY OVERHAULED MAGNETO INSTALLED	250 HOURS		X	2	
2002-19-03		CRANKSHAFT INSP. FOR LYC. 6 CYL. TURBO ENGINES					
553		CRANKSHAFT INSTALLED NOT AFFECTED BY THIS AD	N/A	X		3	
69-08-09		INSTALLATION OF MANIFOLD PRESSURE PLACARD					
N/A		FIELD COMPLIANCE	10 HOURS	X		3	

Codes:

- 1 AD Applicable to Engine
- 2 AD Not Applicable to Engine
- 3 Field Compliance where applicable



Production Automated Engine Test Log

Std Run-in Data Sheet

Order #: AR339795 Preservation: LPS 486

Model #: IO-540-C4D5D Accept/Date : Mchacken Test Mode: Final

10-21-08 Accept/Date :

Part 3.2 Test Sign:

Operator 2: Fuel Serial #: 70D41809

Ignition Left: G08EA039

BOM: HENPL-RT8542

10-22-08

Fuel Slave: NO Fuel Curve: C-12906-E540

Engine #: L-15176-48E

Operator 1: J.McCracken

Ignition Right: Ignit_L Slave : NO Ignit_R Slave : NO Float Bowl Connected to Fuel Pump Inlet: YES

SW Rev : 300-100195 H Barometer: 29.36

Form# T-001

Pre-Oil Temp: 167 ETS #: 642 Pre-Oil Pres: 97

Setting #: 2524054-11

Read Time: 10/21/08 11:50

ETS Rev: EC0-26195 Comments

Date/Time: 10/21/08 11:12 Test Status : Pass

Variations

The following minor variations to the applicable engine test specification were observed during this test and determined to be acceptable 'as is'. The acceptance of these variations will not affect air worthiness or performance.

1) 1.2 149 100 0 1389 AF 1389 AF 2) 3) 4) Engr. Attacherie Q.C. Date 10

When / Test	Low Limit	Value	High Limit	Status
2 Runin @ 1500 for 5 min				
Allowable Speed Variation (RPM)	-75	14.6	75	Pass
Cylinder Head Temperature (DegF)		275.8	500	Pass
Inlet Air Temp (DegF)	0	63.9	110	Pass
5 Runin @ 1800 for 10 min				
Allowable Speed Variation (RPM)	-75	8.4	75	Pass
Cylinder Head Temperature (DegF)		310.8	500	Pass
Inlet Air Temp (DegF)	0	65.3	110	Pass
8 Runin @ 2200 for 10 min				
Allowable Speed Variation (RPM)	-75	6.6	75	Pass
Cylinder Head Temperature (DegF)		373.6	500	Pass
Inlet Air Temp (DegF)	0	66	110	Pass
11 MagChk @ 2200 for 1 min				
Magneto Dropoff - Each Mag (RPM)		117.9	150	Pass
Magneto Dropoff - Difference (RPM)		32.5	35	Pass
Cylinder Head Temperature (DegF)		373.4	500	Pass
Inlet Air Temp (DegF)	0	66.1	110	Pass
15 Idle @ 0 for 4 min				
Idle Speed (RPM)	600	666.1	700	Pass
Cylinder Head Temperature (DegF)		268.5	500	Pass
Oil Pressure @ Idle (PSI)	35	52.1		Pass
Inlet Air Temp (DegF)	0	65.4	110	Pass

Production Automated Engine Test Log

Form# T-001

Std Airflow Data Sheet

Engine #: L-15176-48E

Operator 1: J.McCracken

Operator 2:

Comments

Fuel Serial #: 70D41809

Fuel Curve: C-12906-E540

Fuel Slave: NO

Pre-Oil Temp: 167 ETS #: 642

Model #: 1,0,5440-C4050

Ignition Left: G08EA039

ETS Rev: EC0-26195

Ignit L Slave: NO

Pre-Oil Pres: 97

Accept/Date :

10-21-08

Test Mode: Final

Ignition Right: Ignit R Slave: NO

Float Bowl Connected to Fuel Pump Inlet: YES

Engr.

Setting #: 2524054-11 Date/Time: 10/21/08 12:38

Q.C.

BOM: HENPL-RT8542

Order #: AR339795

SW Rev: 300-100195 H

Barometer: 29.36

Test Status : Review

Date_

Read Time: 10/21/08 13:21

Variations

The following minor variations to the applicable engine test specification were observed during this test and determined to be acceptable 'as is'. The acceptance of these variations will not affect air worthiness or performance. 2) 3) 4)

When / Test	Low Limit	Value	High Limit	Status
2 Airflow @ AF-600 for 4 min		- Incantalia		0.0000000000000000000000000000000000000
Fuel Flow Limits (PPH)	54	56.72	62.0 @ AF-601	Pass
Fuel Nozzle Limits ("HG)	N/A	0.51	N/A	Pass
Cylinder Head Temperature (DegF)		324	500	Pass
Inlet Air Temp (DegF)	0	64.9	110	Pass
4 Airflow @ AF-1000 for 4 min				
Fuel Flow Limits (PPH)	86	87.88	95.0 @ AF-1001	Pass
Fuel Nozzle Limits ("HG)	N/A	3.79	N/A	Pass
Cylinder Head Temperature (DegF)		367.2	500	Pass
Inlet Air Temp (DegF)	0	65	110	Pass
6 Airflow @ AF-1400 for 4 min				viriana de l
Fuel Flow Limits (PPH)	119.9	126.45	131.9 @ AF-1389	Pass
Fuel Nozzle Limits ("HG)	11.2	10	12.3 @ AF-1389	Review
Cylinder Head Temperature (DegF)		400.4	500	Pass
Inlet Air Temp (DegF)	0	64.8	110	Pass
10 Rated @ Rated for 15 min				
Rated Speed (RPM)	2525	2576.6	2625	Pass
Manifold Pressure @ Rated ("HG)	26.5	27.14	28.5	Pass
Fuel Pressure @ Rated (PSI)	18	19.7	28	Pass
Oil Pressure @ Rated (PSI)	75	78.8	85	Pass
Air Flow Limits (PPH)	1464.5	1481.5		Pass
Cylinder Head Temperature (DegF)		414.4	500	Pass
Oil Gallery Temp @ Rated (DegF)	165	167.1	215	Pass
Inlet Air Temp (DegF)	0	65.7	110	Pass
13 Final_Idle @ 0 for 4 min				1.7777
dle Speed (RPM)	600	668	700	Pass
Cylinder Head Temperature (DegF)		301.4	500	Pass
Oil Gallery Temp @ Final_Idle (DegF)	140	174.4	215	Pass
Oil Pressure @ Idle (PSI)	35	53	10-01-00-0	Pass
Manifold Pressure @ Final Idle (In Hg)	8	14.8	17	Pass
Inlet Air Temp (DegF)	0	66.2	110	Pass
15 Manual @ N/A for 0 min				
Acceleration Check (sec)	0	4	5	Pass
Idle Cut-off Time (sec)	0	4	5	Pass

AIRFRAME

FAA Airworthiness Directive Compliance Record

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2000 University Ave. Dubuque, IA 52004 563-589-3812

Content Revision: 4/6/20	D12 File ID: N18	66S	Ai	rcraft Regis	stration: N1866	6S
FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num 2. Cert. Type 4. Author. By
Manufacturer	Category	Model			Part	#: TB-20
Socata Groupe	Airframe	TB 20			Serial	#: 1866
86-17-03 9/22/1986	TO ASSURE THE INTEGRITY OF THE TRIM CONTROL SYSTEM	767.6	NA to aircraft by serial number	Once	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	7		©ATP	Signature.	arnes 5. Jenlevos
86-21-08 10/24/1986	TO PREVENT POSSIBLE STRUCTURAL DAMAGE AND LOSS OF AIRFRAME INTEGRITY	ТАСН	NA to aircraft by serial number	Once	NA	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP	N		©ATP	Signature:	James S. Janlans
87-03-11 3/11/1987	TO PREVENT LOOSE STABILATOR ROD ENDS THAT MAY RESULT IN LOSS OF PITCH CONTROL		NA to aircraft by serial number	Recur	NA	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature.	omes S. Jenhans
87-12-09 R1 6/25/1989	TO PREVENT STRUCTURAL FAILURE OF THE AILERON, POSSIBLE FLUTTER, AND LOSS OF CONTROL	4-10-2012	NA to aircraft by serial number	Recur	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	3-1		©ATP	Signature:	James S. Jenkus
87-22-02 R1 2/26/1990	TO PREVENT STRUCTURAL FAILURE OF THE HORIZONTAL STABILIZER/ ELEVATOR ATTACHMENT AND LOSS OF PITCH CONTROL		NA to aircraft by serial number	Recur	NA O	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	1		©ATP	Signature:	arnes 5. Jenans
88-02-05 2/20/1988	TO DETECT INTERFERENCE BETWEEN THE MOVEABLE PORTIONS OF THE LANDING GEAR STRUCTURE THAT MAY PREVENT EXTENSION, CONTD.		NA to aircraft by serial number	Once	NA /	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP	1		©ATP	Signature.	amo 5. Jenbens
	©ATP	Pri	nted 4/10/2012 7:43:59AM	+	1	ge 1 of 4

2000 University Ave. Dubuque, IA 52004 563-589-3812

Content Revision: 4/6/2	012 File ID: N18	866S	A	ircraft Regi	stration: N186	6S
FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num 2. Cert. Type 4. Author. B
Manufacturer	Category	Model			Part	:#: TB-20
Socata Groupe	Airframe	TB 20			Serial	I#: 1866
90-02-18 R1 6/13/1990	TO PRECLUDE LOSS OF POWER DUE TO CONTAMINATION OF THE FUEL SYSTEM	767.6	NA to aircraft by date of manufacture	Recur	NA	1. University of Dubuque 2. AP 3.3015266 4. James S. Jenkins
©ATP	©ATP	1 7		©ATP	Signature.	arns S. Janker
90-25-17 1/3/1991	TO PREVENT POSSIBLE UNCONTROLLED RELEASE OF FLAMMABLE FLUIDS INTO THE ENGINE COMPARTMENT, CONTD.	ТЭвен	Complied with at date of manufacture	Once	NA)	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	N		©ATP	Signature:	annes S. Jenland
91-05-02 3/25/1991	TO PREVENT STRUCTURAL FAILURE OF THE FUSELAGE FRAME IN THE AREA OF THE LANDING GEAR ATTACHMENT		NA to aircraft by serial number	Recur	NA]	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature:	ames 5. Jenkins
91-12-19 6/20/1991	Superseded by 91-15-10	4-10-13	Superseded by 91-15-10	Once	NA)	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	7		©ATP	Signature:	ames 5. Jenkers
91-15-10 8/10/1991	TO PREVENT ADVERSE AIRPLANE HANDLING QUALITIES AND POSSIBLE LOSS OF CONTROL OF THE AIRPLANE		NA to aircraft by serial number	Once	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature:	ames 5. Jenkens
98-04-03 3/24/1998	TO PREVENT FAILURE OF THE UPPER SEAT BELT ATTACHMENT CAUSED BY EXCESSIVE LOADS ON THE UPPER ATTACHMENT, CONTD.		NA to aircraft by serial number	Once	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	/		©ATP	Signature:	ames 5. Jenkers
	©ATP	Prir	nted 4/10/2012 7:43:59AM		Pag	ge 2 of 4

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2000 University Ave. Dubuque, IA 52004 563-589-3812

Date Time Method of Compliance/Applicability Or Recur Date Time Method of Compliance/Applicability Or Recur Date Time Date Time Date Time Date Da	Content Revision: 4/6/2	012 File ID: N1	866S	Aire	craft Regi	stration: N186	6S
Secala Groupe		Description	Date		or	Date	1. Facility 3. Cert. Num 2. Cert. Type 4. Author. B
98-18-13 10/24/1998 CRACKS IN THE MAIN LANDING CRECK (CRACKS IN THE MAIN LANDING GEAR (MLG) ATTACHMENT BEARING, WHICH COULD RESULT IN, CONTD. @ATP 2001-23-04 1/4/2002 To detect & correct fatigue cracks in the lower rudder hinge fitting @ATP 2001-23-05 1/4/2002 To eliminate the potential for the front seats to inadvertently unlock from their fixed positions, contd. @ATP 2003-04-03 4/7/2003 To prevent failure of the aileron control gimbal joint @ATP 2007-02-04 2/26/2007 @ATP @ATP	Manufacturer	Category	Model			Part	#: TB-20
CRACKS IN THE MAIN LANDING GEAR (MLG) ATTACHMENT BEARING, WHICH COULD RESULT IN, CONTD. ©ATP 2001-23-04 1/4/2002 To detect & correct fatigue cracks in the lower rudder hinge fitting ©ATP 201-23-05 1/4/2002 To eliminate the potential for the front seats to inadvertently unlock from their fixed positions, contd. ©ATP 203-04-03 4/7/2003 To prevent failure of the aileron control gimbal joint ©ATP 2007-02-04 2/26/2007 The MCAI states there are reports of interference between the wing spar lower boom and the wheel, contd. ©ATP Once Tick 1. Universit 2. AP 3. 3015266 4. James S CAP 3. 3015266 4. J	Socata Groupe	Airframe	TB 20			Serial	#: 1866
2001-23-04 1/4/2002 In the lower rudder hinge fitting SEE RECURING AD LIST SEE RECURRING AD LIST SIgnature: SEE RECURRING AD LIST Once 1/4/2002 SIgnature: SEE RECURRING AD LIST SIgnature: SEE RECURRING AD LIST PCW by Socata S/B 10-115-25 SEE #1 AF LOG 1-13-04, tach 423.3 SEE RECURRING AD LIST SEE RECURRING AD LIST SIgnature: SEE RECURRING AD LIST SIgnature: SEE RECURRING AD LIST SIgnature: SEE RECURRING AD LIST SIgnature: SEE RECURRING AD LIST AD LIST Signature: SEE RECURRING A		CRACKS IN THE MAIN LANDING GEAR (MLG) ATTACHMENT BEARING, WHICH COULD	RECURING	SEE RECURRING AD LIST	Recur	RECURING	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
SEE RECURING AD LIST SEE RECURING AD LIST SEE RECURING AD LIST Signature: SEE RECURING AD LIST Sig	©ATP	©ATP			©ATP		SEE RECURRING AD LIST
To eliminate the potential for the front seats to inadvertently unlock from their fixed positions, contd. ©ATP 2003-04-03 4/7/2003 ©ATP ©ATP Control gimbal joint ©ATP Control gimbal joint ©ATP ©ATP 2007-02-04 2/26/2007 ©ATP PCW by relocate screw holes per para (d) of SB 10-148. SEE #1 AF LOG 3-12-07, maint hobbs 198.3 ©ATP ©ATP ©ATP Once 1. University 2. AP 3. 3015266 4. James S Signature: PCW by relocate screw holes per para (d) of SB 10-148. SEE #1 AF LOG 3-12-07, maint hobbs 198.3 Signature: ©ATP Signature: SEE RECURING AD LIST 3. 3015266 4. James S Signature: Once NA 1. University 2. AP 3. 3015266 4. James S Signature: Once ATP Signature: Once 1. James S Signature: Once 3. 3015266 4. James S Signature: Once 1. James S Signature: Once 3. 3015266 4. James S Signature: Once 1. James S Signature: Once 3. 3015266 4. James S Signature: Once 3. 3015266 4. James S Signature: Once 1. James S Signature: Once 1. James S Signature: Once 1. James S Signature: Once Once 1. James S Signature: Once	***************************************		RECURING	SEE RECURRING AD LIST	Recur	RECURING	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
front seats to inadvertently unlock from their fixed positions, contd. ### Factor	©ATP	©ATP			©ATP	Signature: S	SEE RECURRING AD LIST
©ATP 2003-04-03 4/7/2003 To prevent failure of the aileron control gimbal joint SEE RECURING AD LIST ©ATP ©ATP ©ATP ©ATP SEE RECURING AD LIST ©ATP SEE RECURING AD LIST ©ATP Signature: SEE RECURING AD LIST 3.3015266 4. James Signature: PCW by relocate screw holes per para (d) of SB 10-148. SEE #1 AF LOG 3-12-07, maint hobbs 198.3 ©ATP Signature: ©ATP Signature: ©ATP Signature: SER RECURING AD LIST 3.3015266 4. James Signature: Once Once TA CH 17 6 7. 6 Signature: Once ATP Signature: Once 1. 2. 3. 3. Signature: Once ATP Signature: Once		front seats to inadvertently unlock	TACH		Once		1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
2003-04-03 4/7/2003 To prevent failure of the aileron control gimbal joint SEE RECURING AD LIST Signature: SEE RECURING AD LIST Once A James S Signature: SEE RECURING AD LIST Once A James S SIGNATURE: SEE RECURING AD LIST Once A James S SIGNATURE: SEE RECURING AD LIST A James S SIGNATURE: SEE RECURING AD LIST Once A James S SIGNATURE: SEE RECURING AD LIST Once A James S SIGNATURE: SEE RECURING AD LIST Signature: SEE RECURING AD LIST A James S SIGNATURE:	©ATP	©ATP			©ATP	Signature:	1
©ATP ©ATP The MCAI states there are reports of interference between the wing spar lower boom and the wheel,contd. ©ATP The MCAI states there are reports of interference between the wing spar lower boom and the wheel,contd. ©ATP PCW by relocate screw holes per para (d) of SB 10-148. SEE #1 AF LOG 3-12-07, maint hobbs 198.3 ©ATP ©ATP Signature: SEE RECUR 7. A C. H	2003-04-03		RECURING	SEE RECURRING AD LIST		SEE) RECURING	1. University of Dubuque
The MCAI states there are reports of interference between the wing spar lower boom and the wheel, contd. The MCAI states there are reports of interference between the wing spar lower boom and the wheel, contd. The MCAI states there are reports of interference between the wing spar lower boom and the wheel, contd. The MCAI states there are reports of interference between the wing spar lower boom and the wheel, contd. PCW by relocate screw holes per para (d) of SB 10-148. SEE #1 AF LOG 3-12-07, maint hobbs 198.3 Signature: Once NA 1. University 2. AP 3. 3015266 4. James S Once Once NA 1. University 2. AP 3. 3015266 4. James S Once 3. 3015266 4. James S Once 1. 2. 3.	©ATP	©ATP			©ATP	Signature: S	SEE RECURRING AD LIST
©ATP		of interference between the wing spar lower boom and the	TACH	SB 10-148. SEE #1 AF LOG 3-12-07,			1. University of Dubuque
9 Once 1. 2. 3.	©ATP	©ATP			©ATP	Signature:	ames 5. Jenkins
Signature:	9				Once		1. 2. 3.
©ATP Printed 4/10/2012 7:43:59AM Page 3 of 4							JOHN CANTON

IQ-540-C4B5D Engine Model: Authorized Signiture

AD & REV#	REV DATE	APPLIC. SB #	ONETIME	RECURRING
92-12-05	To the second se	501B	/ _x	
95-07-01		95-002	X	
96-09-10		456F 524	X	
97-15-11		527C	x	
98-17-11			X	
2002-12-07		543A		X
2004-05-24		554	X	
2004-10-14		475C	X	
2005-19-11		566	X	

AIRWORTHINESS DIRECTIVE repair

At date of engine XXXXXX:

08/14/2012

Engine Model:

	Authorized Signiture	: Atm	-119	
AD & REV#	REV DATE	APPLIC. SB #	ONE TIME	RECURRING
2006-20-09	-	569A	x//	
2009-02-03		PRS107 R4	X X	
2011-26-04		342E		x
82-12-06 R1	11/12/81	618 619	. X	
82-20-01		623	X	
2005-12-06		MSB645		X
END				

C	OMPLIANCE RECORD
S/N _ L-15176,-48E	W/O 68818
C/W complied with, P/C/	Mpreviously complied with, N/A not applicable
CRS MW2R020L	-1111
1	
	N/A/by P/N
	C/W new lycoming rod bolts installed
And the second s	P/C/W
	N/A by P/N
	C/W crankshaft inspected
next overhaul	P/C/W
	N/A by model
	C/W crank & gear inspected
	N/A by S/N

COMPLI	ANCE RECORD
S/NL-15176-48E	W/O _68818
C/W complied with, P/C/W previo	ously complied with, N/A not applicable
	N/A by S/N
	P/C/W
next 100 hour annual overhaul any fuel line maint.	C/W fuel inj lines clamps inspected
	P/C/W
	P/C/W
500 hours	C/W cam assy checked ok
END	

ENGINE

FAA Airworthiness Directive Compliance Record

2000 University Ave. Dubuque, IA 52004 563-589-3812

Content Revision: 4/6/2	012 File ID: N18		Ai	rcraft Regi	stration: N186	6S
FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Nun 2. Cert. Type 4. Author. B
Vlanufacturer	Category	Model			Part	#: IO-540-C4D5D
Textron Lycoming	Engine	IO-540-C4D5	ip .		Serial	#: L-15176-48E
63-14-03 7/6/1963	OIL PUMP DRIVE SHAFT	267.6	NA by engine serial number	Recur	NA	1. University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP	1 7		©ATP	Signature:	amas S. Jandios
66-20-04 8/27/1966	TO PREVENT FURTHER FAILURES OF OIL FILTER ADAPTER GASKET, P/N 74904	ТАСН	NA by engine serial number	Once	NA (1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature.	ames 5. Jenlando
75-08-09 R(3) 8/18/1977	TO PREVENT OIL PUMP FAILURES, INSPECT, REPLACE AND ASSEMBLE THE OIL PUMP DRIVE SHAFT AND DRIVE IMPELLER		NA by engine serial number	Once	NA]	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP	/		©ATP	Signature:	
75-09-15 4/30/1975	TO PREVENT POSSIBLE FUEL STARVATION TO THE ENGINE	4-10-13	NA by engine serial number	Once	NA I	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP TO PREVENT AN IN-FLIGHT	1		©ATP	Signature:	James S. Leubins
78-23-10 11/7/1978	POWER LOSS DUE TO AN OVER RICH CONDITION, CONTD.		NA by parts list number	Once	NA	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature:	ames S. Jenams
79-04-05 9/26/1979	TO PREVENT AN IN-FLIGHT POWER LOSS DUE TO THE SEPARATION OF THE P/N 2529192 REGULATOR DIAPHRAGM STEM ASSEMBLY		NA by parts list number	Once	NA T	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP	I		©ATP	Signature:	anno S. Jenberos
	©ATP	Prir	nted 4/10/2012 7:42:32AM		1	1 of 12

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2000 University Ave. Dubuque, IA 52004

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Content Revision: 4/6/20	D12 File ID: N1	866S	Ai	rcraft Regi	stration: N186	6S
FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num 2. Cert. Type 4. Author. B
V anufacturer	Category	Model			Part	:#: IO-540-C4D5D
Textron Lycoming	Engine	IO-540-C4D5	D		Serial	I#: L-15176-48E
81-18-04 R2 6/7/1982	Superseded by 96-09-10	767.6	Superseded by 96-09-10	Once	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature:	amos Dendus
87-10-06 R1 9/1/1989	TO PREVENT POSSIBLE ROCKER ARM FAILURE AND LOSS OF ENGINE POWER, INSPECT AND REWORK OR REPLACE ROCKER ARM,	ТАСН	NA by engine serial number	Once	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	CONTD. ©ATP	N		©ATP	Signature:	arness. Jensons
91-14-22 8/19/1991	Superseded by 2004-10-14		Superseded by 2004-10-14	Recur	NA I	1. University of Dubuque 2. AP 3. 3015266
©ATP	©ATP			©ATP	Signature:	4. James S. Jenkins
92-12-05 7/10/1992	TO PREVENT PISTON PIN FAILURE, OR PISTON RELEASE, AND ENGINE FAILURE	4-10-13	NA by engine serial number	Once	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	7		©ATP	Signature:	0 0
93-02-05 6/14/1993	Superseded by 2002-26-01		Superseded by 2002-26-01	Recur	NA NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature:	ames 5. Jenkus
94-14-13 L 6/23/1994	Superseded by 95-26-02		Superseded by 95-26-02	Once	NA (1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
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FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num 2. Cert. Type 4. Author. B
Manufacturer	Category	Model			Part	#: IO-540-C4D5D
Textron Lycoming	Engine	IO-540-C4D5	5D		Serial	#: L-15176-48E
95-07-01 4/12/1995	TO PREVENT ENGINE FAILURE DUE TO CONNECTING ROD BOLT FAILURE, WHICH COULD RESULT IN DAMAGE TO OR LOSS,CONTD.	767.6	NA no "Superior Air Parts" parts installed	Once	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	Y		©ATP	Signature:	answa S. Jackina
95-26-02 1/24/1996	TO PREVENT DETONATION DUE TO LOW OCTANE, WHICH CAN RESULT IN SEVERE ENGINE DAMAGE AND SUBSEQUENT FAILURE	ТАЕН	NA by aircraft registration	Once	NA T	1. University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP	H		©ATP	Signature: Q	ames S. Jenlins
96-09-10 C 7/15/1996	TO PREVENT OIL PUMP FAILURE DUE TO IMPELLER FAILURE, WHICH COULD RESULT IN AN ENGINE FAILURE		NA to eninge by model (-C4D5D)	Once	NA J	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP	/		©ATP	Signature:	amos Jeulas
96-23-03 12/17/1996	TO PREVENT AN INFLIGHT ENGINE FAILURE DUE TO FUEL STARVATION, WHICH COULD RESULT IN A FORCED LANDING	4-10-13	NA by date of manufacture	Once	NA J	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	N		©ATP	Signature: 4	ames S. Jenlaus
97-01-03 1/21/1997	Superseded by 97-15-11		Superseded by 97-15-11	Once	NA /	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature:	arnes S. Teuloun
97-15-11 8/12/1997	TO PREVENT PISTON PIN FAILURE, WHICH COULD RESULT IN ENGINE FAILURE		NA by engine serial number	Once	NA J	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature:	
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FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Nun 2. Cert. Type 4. Author. B
Manufacturer	Category	Model			Part	#: IO-540-C4D5D
Textron Lycoming	Engine	10-540-C4D5	D		Serial	#: L-15176-48E
98-17-11 C 10/19/1998 ©ATP	TO PREVENT CRANKSHAFT FAILURE DUE TO CRACKING, WHICH COULD RESULT IN AN INFLIGHT ENGINE FAILURE AND POSSIBLE, CONTD. ©ATP	1767.6	PCW @ OH. SEE Lycoming AD LIST	Once	NA Ciarata Ch	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
98-18-12	Superseded by 2003-14-03	-		©ATP	Signature:	annes S. Cantains
9/28/1998	- Caponedada by 2000 14 00	Нэв	Superseded by 2003-14-03	Recur	NA /	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	N		©ATP	Signature:	- Carries C. Schalls
2000-18-53 E 9/5/2000	Superseded by 2002-12-07		Superseded by 2002-12-07	Recur	NA T	1. University of Dubuque 2. AP 3.3015266
©ATP	©ATP			©ATP	Signature:	4. James S. Jenkins
2002-12-07 7/3/2002	To prevent complete loss of engine oil and subsequent seizing of the engine and possibility of fire,contd.	4-10-12	PCW at OH. New PN converter plate installed. SEE Lycoming AD LIST	Recur	NA	1.University of Dubuque 2.AP 3.3015266
©ATP	©ATP	N.		©ATP	Signature:	4. James S. Jenkins
2002-19-03 9/20/2002	To prevent crankshaft failure, which could result in total engine power loss, in-flight engine failure and,contd.		NA to engine - no turbo charger installed	Once	NA)	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature:	N 1
2002-20-51 E 10/1/2002	Superseded by 2002-23-06		Superseded by 2002-23-06	Once	NA T	1. University of Dubuque 2. AP 3.3015266
©ATP	©ATP			©ATP	Signature:	4. James S. Jenkins
	5 Section 25 4 555			WAIP	Signature.	ernes S. Jentero
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Content Revision: 4/6/2	012 File ID: N1	866S	A	ircraft Regi	stration: N186	6S
FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num 2. Cert. Type 4. Author. By
Manufacturer	Category	Model			Part	:#: IO-540-C4D5D
Textron Lycoming	Engine	IO-540-C4D5	SD .		Serial	#: L-15176-48E
2002-23-06 11/19/2002	Superseded by 2004-05-24	267.6	Superseded by 2004-05-24	Once	NA	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP	1 7		©ATP	Signature:\(\)	
2002-26-01 1/31/2003	To prevent failure of the fuel injector fuel lines allowing fuel to spray into the engine compartment, resulting, contd.	Таси	Superseded by 2008-14-07	Recur	NA NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	١ '٢		©ATP	Signature:	James 5. Denlar
2003-14-03 8/14/2003	To prevent rotary fuel pump leaks, which could result in an engine failure, engine fire, and damage to or,contd.	4-10-13	NA by fuel pump type / PN	Recur	NA /	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP	1 34		©ATP	Signature:	amen S. Jewasons
2004-05-24 C 3/30/2004	To prevent the loss of all engine power and possible forced landing		NA to enigne per para (c) (4)	Once	NA NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	/		©ATP	Signature:	emis 5. Jeanless
2004-10-14 C 6/25/2004	To prevent loosening or failure of the crankshaft gear retaining bolt, which may cause sudden engine failure		DUE AT PROPELLER STRIKE OR OVERHAUL (OH)	Once	ON CONDITION	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	/		©ATP	Signature:	emes. Jentins
2005-12-06 7/19/2005	To prevent failure of the magneto impulse coupling assembly and possible engine failure	SEE RECURING AD LIST	SEE RECURRING AD LIST PCW @ OH	Recur	SEE TRECURING AD LIST	1.Universitylof Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature: S	SEE RECURRING AD LIST
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Content Revision: 4/6/20	012 File ID: N18	866S	Aire	craft Regi	stration: N1866	SS
FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num 2. Cert. Type 4. Author. B
Manufacturer	Category	Model			Part	#: IO-540-C4D5D
Textron Lycoming	Engine	10-540-C4D5	5D		Serial	#: L-15176-48E
2005-19-11 10/21/2005	To prevent failure of the crankshaft, which could result in total engine power loss, in-flight failure, and,contd.	7626	PCW at OH. Replaced crankshaft. SEE Lycoming AD LIST	Once	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature:	ams 5. Jenkers
2005-26-10 1/31/2006	Superseded by 2006-12-07	ТАСН	Superseded by 2006-12-07	Once	NA	1. University of Dubuque 2. AP 3. 3015266
©ATP	©ATP	1 17		©ATP	Signature.	4. James S. Jenkins
2006-10-21 C2 6/22/2006	To prevent fatigue failure of the connecting rod & possible uncommanded shutdown of the engine		NA to engine. No - ECi parts installed	Once	NA NA	1. University of Dubuque 2. AP 3.3015266
©ATP	©ATP			©ATP	Signature:	4. James S. Jenkins
2006-12-07 7/11/2006	To prevent loss of engine power due to cracks in the cylinder assemblies & possible engine failure caused, contd.	4-10-13	NA to engine. No - ECi parts installed	Once	NA T	1. University of Dubuque 2. AP 3.3015266 4. James S. Jenkins
©ATP	©ATP	2		©ATP	Signature:	4. James S. Jenkins
2006-20-09 11/3/2006	To prevent failure of the crankshaft, which will result in total engine power loss, in-flight engine,contd.		PCW at OH. Replaced crankshaft. SEE Lycoming AD LIST 米 SUPER SEOED しょ 2013-19-01米	Once	NA NA	1. University of Dubuque 2. AP 3.3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature:	ame 5. July
2007-04-19 R1 5/7/2007	To prevent cylinder separation that can lead to engine failure, possible engine compartment fire, and,contd.		NA to engine. NO - Superior Air Parts (SAP) clylinder assemblies installed	Once	NA)	1.University of Dubuque 2.AP 3.3015266
©ATP	©ATP			©ATP	Signature:	4. James S. Jenkins
	©ATP	Prin	nted 4/10/2012 7:42:32AM	OATT	1	6 of 12

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2000 University Ave. Dubuque, IA 52004

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Content Revision: 4/6/2	012 File ID: N1	866S	A	ircraft Regi	stration: N186	6S
FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Nun 2. Cert. Type 4. Author. B
Manufacturer	Category	Model			Part	:#: IO-540-C4D5D
Textron Lycoming	Engine	IO-540-C4D5	5D		Serial	I#: L-15176-48E
2008-06-51 E 3/12/2008	Superseded by 2008-08-14	767.6	Superseded by 2008-08-14	Recur	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	1		©ATP	Signature:	ames S. Jenkens
2008-08-14 4/29/2008	To prevent a lean running engine, which could result in a substantial loss of engine power and substantial,contd.	ТАСН	Superseded by 2009-02-03	Recur	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	M		©ATP	Signature:	
2008-14-07 8/14/2008	To prevent failure of the fuel injector fuel lines that would allow fuel to spray into the engine compartment, contd.		Superseded by 2011-26-04	Recur	NA NA	1. University of Dubuque 2. AP 3. 3015266
©ATP	©ATP			©ATP	Signature:	4. James S. Jenkins
2008-19-05 10/20/2008	To prevent loss of engine power due to cracks at the head-to-barrel interface in the cylinder assemblies, contd.	4-10-13	Superseded by 2009-26-12	Recur	NA /	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	34		©ATP	Signature: ()	4. James S. Jenkins
2009-02-03 2/9/2009	To prevent a lean running engine, which could result in a substantial loss of engine power and subsequent, contd.		NA has "G" stamped on hex plug	Recur	NA /	1. University of Dubuque 2. AP 3.3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature:	anes 5. Jens
2009-26-12 2/4/2010	To prevent loss of engine power due to cracks at the head-to-barrel interface and possible engine failure,contd.		NA to engine. No - ECi parts installed	Recur	NA)	1.University of Dubuque 2.AP 3.3015266
©ATP	©ATP			©ATP	Signature:	4. James S. Jenkins
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	©ATP	Prin	ited 4/10/2012 7:42:32AM		Page	7 of 12

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FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num 2. Cert. Type 4. Author. By
Manufacturer	Category	Model			Part	#: IO-540-C4D5D
Textron Lycoming	Engine	10-540-C4D5	5D		Serial	I#: L-15176-48E
2011-15-10 8/16/2011	To correct an AFS fuel servo diaphragm	4-10-12 TACH 1767.6	Superseded by 2012-03-06C	Once	NA	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature.	James S. Jenlass
2011-26-04 1/25/2012	To prevent failure of the fuel injector fuel lines that would allow fuel to spray into the engine compartment, contd.	SEE RECURING AD LIST	SEE RECURRING AD LIST	Recur	SEE RECURING AD LIST	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature: S	SEE RECURRING AD LIST
2012-03-06 C 2/24/2012	To prevent an in-flight engine shutdown due to a failed fuel servo diaphragm, and damage to the airplane	25.3	NA to fuel servo by serial number IAW AVStar AFS-SB6 and Lycoming SB-596	Once	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	/		©ATP	Signature.	4.02
2012-03-07 3/27/2012	To prevent engine in-flight shutdown, power loss, and reduced control of the airplane	37	NA HA-6 carberuator NOT installed	Once	NA V	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	/		©ATP	Signature:	ones 5. Jenles
2012-19-01 10/24/2012	To prevent filure of the Crankshaft, which will result in total engine dower loss, in-flight engine	2-7-2013 M214 HOBBS 195414	NA to engine serial mumber	Once	NA	1. University of Dubuque 2. Ap 3. 3015 2 66 4. Tames S. Jenk INS umes S. Denkins
9				Once	Signature:	1. 2. 3. 4.
	ea.TD	<u></u>			1	The colores
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Report Produced By: James Jenkins

Content Revision: 4/6/2012 File ID: N1866S Aircraft Registration: N1866S Complied Amendment Number **FAA AD Number** Once **Next Due** 1. Facility 3. Cert. Num. Description Date **Effective Date** Date Method of Compliance/Applicability or 2. Cert. Type 4. Author. By Time Recur Time Manufacturer Category Model Part #: Fuel Injected System Precision Airmotive RSA-5AD1 Serial #: TO DETECT DEFECTIVE 73-10-02 Recur NA 1. University of Dubuque DIAPHRAGM ASSEMBLIES 5/16/1973 NA - due to last OH date 2.AP 3.3015266 4. James S. Jenkins ©ATP CATP **©ATP** Signature. mes 5 TO PREVENT A FUEL FLOW 79-21-08 Once 1. University of Dubuque CUTOFF TO THE ENGINE AND 10/24/1979 NA to fuel servo by part number 2.AP SUBSEQUENT LOSS OF POWER 7ACH 3.3015266 4. James S. Jenkins **©ATP** ©ATP **©ATP** Signature: REGULATOR STEM AND LOCK 79-26-03 Once NA 1. University of Dubuque 12/26/1979 NA to fuel servo by part number 2.AP 3.3015266 4. James S. Jenkins **©ATP** ©ATP **©ATP** Signature:\ Superseded by 2008-08-14 2008-06-51 E Recur NA 1. University of Dubuque 4-10-13 3/12/2008 Supersded by 2008-08-14 2.AP 3.3015266 4. James S. Jenkins **©ATP** ©ATP **©ATP** Signature: 2008-08-14 To prevent a lean running engine, Recur NA 1. University of Dubuque which could result in a substantial 4/29/2008 Superseded by 2009-02-03 2.AP loss of engine power and 3.3015266 substantial,contd. 4. James & Jenkins **©ATP** ©ATP **©ATP** Signature: To prevent a lean running engine, 2009-02-03 Recur NA 1. University of Dubuque which could result in a substantial 2/9/2009 NA not rebuilt, overhauled, or repaired since 2.AP loss of engine power and August 22, 2006. 3.3015266 subsequent, contd. 4. James S. Jenkins **©ATP ©ATP** ©ATP Signature: **©ATP** Printed 4/10/2012 7:42:32AM Page 10 of 12

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Content Revision: 4/6/2	012 File ID: N1	366S	Ai	rcraft Regi	stration: N186	6S
FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num 2. Cert. Type 4. Author. By
Manufacturer	Category	Model			Part	#:
Precision Airmotive	Fuel Injected System	RSA-5AD1			Serial	#:
2012-03-06 C 2/24/2012	To prevent an in-flight engine shutdown due to a failed fuel servo diaphragm, and damage to the airplane	4-10-12 TACH 1767.6	NA to fuel servo by serial number IAW AVStar AFS-SB6 and Lycoming SB-596	Once		1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature	h 5 0 a ll
9				Once		1. 2. 3. 4.
					Signature:	
Manufacturer	Category	Model			Part	#:
Textron Lycoming	Fuel Pumps	LW15473			Serial	#:
92-20-07 L 10/5/1992	Superseded by 93-05-21	74.5	Superseded by 93-05-21	Once	NA	1. University of Dubuque 2. AP 3. 3015266
©ATP	©ATP	1 7		©ATP	Signature:	4. James S. Jenkins
93-05-21 L 3/25/1993	Superseded by 93-11-11	4.10-12	Superseded by 93-11-11	Once	NA)	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP	×		©ATP	Signature	4. James S. Jenkins
93-11-11 6/21/1993	TO PREVENT DISRUPTION OF FUEL FLOW TO THE ENGINE, WHICH CAN RESULT IN A LOSS OF ENGINE POWER		SEE Lycoming ENG AD LIST NA by diaphragm fuel pump part number and new pump installed at time of engine OH. New AC F/P inst. when applic.	Once	NA /	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP	/		©ATP	Signature:	ames S. Jenles
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FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num 2. Cert. Type 4. Author. By
Manufacturer	Category	Model			Part	#: BL-682560-11
Bendix Corporation	Magnetos	D-3000 SER	IES		Serial	#: G02JA156
78-09-07 R3 1/17/1983	Superseded by 96-12-07	767.6	Superseded by 96-12-07	Recur	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	1 4		©ATP	Signature:	Comes 5. Jenks
81-12-06 R1 11/12/1981	TO DETECT LOOSE DISTRIBUTOR GEAR ROTATING ELECTRODES ON MAGNETOS	Zdun.	NA by magneto part number.	Recur	9	1. University of Dubuque 2. AP 3. 3015266
©ATP	©ATP			©ATP	Signature:	4. James S. Jenkins
82-20-01	TO PREVENT FAILURE OF			Once	NA NA	1. University of Dubuque
6/14/1983 ©ATP	IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD.	4-10-13	NA by serial number			2.AP 3.3015266 4. James S. Jenkins
	©ATP	1		©ATP		ans 5. Jenkos
96-12-07 7/18/1996	Superseded by 2005-12-06		Superseded by 2005-12-06	Recur	NA	1.University of Dubuque 2.AP 3.3015266
©ATP	©ATP	/		©ATP	Signature:	4. James S. Jenkins
2005-12-06 7/19/2005	To prevent failure of the magneto impulse coupling assembly and possible engine failure	SEE RECURING AD LIST	SEE RECURRING AD LIST PCW @ OH	Recur	SEE RECURING AD LIST	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature: S	EE RECRRING AD LIST

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Content Revision: 4/6/2	012 File ID:	N1866S	Ai	ircraft Registration: N1866S			
FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num 2. Cert. Type 4. Author. By	
Manufacturer	Category	Model			Part	#: HC-C2YK-1BF/F8477-4	
Hartzell Propeller	Propeller	HC-C2YK-1			Serial	I#: NS153B	
64-20-01 8/28/1964	PLASTIC PITCH CHANGE BLOCKS	767.6	NA to propeller by model number (-1BF)	Recur	NA	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins	
©ATP	©ATP	1		©ATP	Signature:	ans 5. Jehn	
70-02-01 1/1/1970	Superseded by 73-10-03	TALK	Superseded by 73-10-03	Once	NA	1.University of Dubuque 2.AP 3.3015266	
©ATP	©ATP			©ATP	Signature.	4. James S. Jenkins	
70-16-03 R 1/1/1970	Superseded by 77-12-06		Superseded by 77-12-06	Once	NA NA	1. University of Dubuque 2. AP 3. 3015266	
©ATP	©ATP			©ATP	Signature.	4. James S. Jenkins	
73-10-03 1/1/1973	Superseded by 77-12-06	4-10-13	Superseded by 77-12-06	Once	NA NA	1. University of Dubuque 2. AP 3. 3015266	
©ATP	©ATP	2		©ATP	Signature:	4. James S. Jenkins	
74-15-02 1/1/1974	Superseded by 77-12-06		Superseded by 77-12-06	Once	NA /	1. University of Dubuque 2. AP 3. 3015266	
©ATP	©ATP	1/		©ATP	Signature	4. James S Jenkins	
75-07-05 5/1/1977	Superseded by 77-12-06		Superseded by 77-12-06	Once	NA /	1. University of Dubuque 2. AP 3. 3015266	
©ATP	©ATP	/		©ATP	Signature:	4. James S. Jenkins	
	©ATP	T.		e/Air	oignature.	James S. Jenkins	

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FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Nun 2. Cert. Type 4. Author. B
Manufacturer	Category	Model			Part	#: HC-C2YK-1BF/F8477-4
Hartzell Propeller	Propeller	HC-C2YK-1			Serial	#: NS153B
77-12-06 R(2) 12/21/1977	Superseded by 2002-09-08	767.6	Superseded by 2002-09-08	Recur	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	1		©ATP	Signature:	James 5 Clarke
90-02-23 L 2/5/1990	Superseded by 2001-23-08	ТАСМ	Superseded by 2001-23-08	Recur	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	N		©ATP	Signature:	ames 5. Jenles
2001-07-03 C 6/4/2001	To prevent propeller failure of the propellers returned to service by BASCO, & possible loss of airplane control		NA to propeller by not haveing ever been to "BASCO" repair facility or by serial number	Once	NA	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature:	ans 5: Sentins
2001-23-08 12/24/2001	To prevent failure of the propeller hub resulting from cracks, that can cause blade separation & subsequent, contd.	4-10-13	NA to propeller by aircraft installation also NA by engine horsepower rating (not over 300)	Recur	NA NA	1. University of Dubuque 2. AP 3.3015266 4. James S. Jenkins
©ATP	©ATP	*/		©ATP	Signature:	ames S. Jalle
2002-09-08 6/13/2002	To prevent failure of the propeller blade from fatigue cracks in the blade shank radius, which can,contd.		Superseded by 2007-26-09	Once	NA	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature:	ans 5 Jeulans
2003-01-03 1/23/2003	To prevent in-flight propeller blade separation resulting in airframe and engine damage, & possible loss of the airplane		NA to propeller by hub serial number	Once	NA /	1.University of Dubuque 2.AP 3.3015266
©ATP	©ATP			©ATP	Signature:	4. James S. Jenkins
		7		UAII.	Signature.	anso . Jews
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Content Revision: 4/6/2	012 File ID: N1	866S	Airc	raft Regi	stration: N186	6S
FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num 2. Cert. Type 4. Author. By
Manufacturer	Category	Model			Part	#: HC-C2YK-1BF/F8477-4
Hartzell Propeller	Propeller	HC-C2YK-1			Serial	#: NS153B
2003-06-02 4/29/2003	To prevent propeller blade separation, damage to the airplane, and possible loss of the airplane	267.6	Propeller NEW and deice boots installed by Aircraft Propeller Services Inc, IAW Hartzel maint manuals	Once	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	7		©ATP	Signature:	The same of the same
2003-13-17 7/18/2003	To detect unsafe conditions that could result in separation of a propeller blade & loss of control,contd.	<i>ТАсн</i>	NA to propeller by prop never being serviced by T and W Propellers, Inc.	Once	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	N		©ATP	Signature:	- Common of the
2005-14-11 8/17/2005	To prevent blade failure that could result in separation of a propeller blade and loss of control of the airplane		NA to propeller by prop never being serviced by by Southern California Propeller Service of Inglewood, CA.	Once	NA	1. University of Dubuque 2. AP 3. 3015266 4. James Ş. Jenkins
©ATP	©ATP			©ATP	Signature.	- Connect of Connection
2006-18-15 9/25/2006 ©ATP	To prevent failure of the propeller hub causing blade separation and subsequent loss of airplane control	4-10-13	Superseded by 2009-22-03	Recur	NA /	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
2006-24-07	To detect potentially unsafe			©ATP	Signature:	annes S. Jech
1/3/2007	conditions that could result in a propeller blade separating from the hub, contd.		NA to propeller by never being serviced by Oxford Aviation Limited, doing business as CSE Aviation	Once	NA	1. University of Dubuque 2. AP 3.3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature:	2000
2007-26-09 1/30/2008	To prevent failure of the propeller blade from fatigue cracks in the aluminum blade shank radius, which can,contd.		NA to propeller by blade serial numbers	Once	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
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FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility	3. Cert. Num 4. Author. B
Manufacturer	Category	Model		•	Part	#: HC-C2YK-	1BF/F8477-4
Hartzell Propeller Propeller HC-C2YK-1							
2009-22-03 11/12/2009	To prevent failure of the propeller hub causing blade separation and subsequent loss of airplane control	4-10-12 TACH 1767.6	NA to propeller by engine model installed on	Recur	NA	1. University o 2. AP 3. 3015266	is
©ATP	©ATP			©ATP	Signature.	4. James S. de	enkins
9				Once	Signature.	1.	Jenker
160					Signature:	2. 3. 4.	
Manufacturer	Category	Model			Part	#:	
Induction Air Filters	Air Filter	Any Model			Serial		
84-26-02 1/29/1985	TO PREVENT POSSIBLE ENGINE POWER LOSS OR STOPPAGE CAUSED BY ENGINE INGESTION OF FRAGMENTS, CONTD.	4-10-12 TACH 1767.6	NA to aircraft. Paper induction filter not installed	Recur		1. University of 2. AP 3. 3015266	
©ATP	©ATP			©ATP	Signature:	4. James S. Je	nkins

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FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num 2. Cert. Type 4. Author. By
Manufacturer	Category	Model			Part	
Any Manufacturer	Alternators				Serial	#:
72-01-05 1/1/1972	Superseded by 72-15-02	7464	Superseded by 72-15-02	Once	NA C	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	<i>y</i>		©ATP	Signature:	James S. Jenkins
72-15-02 9/1/1972	TO PRECLUDE IN-SERVICE FAILURES OF ALTERNATOR COOLING FANS	4.10-12	NA to alternator by fan part number	Recur	NA	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins
©ATP	©ATP	37		©ATP	Signature:	ames 5. Jews
76-02-07 2/2/1976	TO DETECT DEFECTIVE ALTERNATOR SLIP RING END BEARINGS AND MINIMIZE THE PROBABILITY OF IN-SERVICE FAILURES		NA to alternator by model number installed	Recur	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins
©ATP	©ATP	/		©ATP	Signature.	James S. Jenker
Manufacturer	Category	Model			Part	
Garmin International	GPS/NAV/COM	GNS 430			Serial	#:
2001-23-17 12/28/2001	To prevent external noise from causing inaccurate course deviation displays in the GNS 430 unit's course, contd.	4-10-12 TACH 1767.6	NA to GNS-430 by part number	Once	NA	1. University of Dubuque 2. AP 3.3015266 4. James S. Jenkins
©ATP	©ATP			©ATP	Signature:	anna 5. Jenlans
Manufacturer	Category	Model		Color Conse	Part	
Anjou Aeronautique	Safety Belts	Any Model			Serial	#:
2003-26-06 2/17/2004	To detect and correct defective safety belts and restraint systems, which could result in failure of,contd.	4-10-12 TACH 1767.6	NA in front by buckle type. NA to rear by manufacture - Pacific Scientific and AM Safe	Recur	NA C	1.University of Dubuque 2.AP 3.3015266 4. James Ş. Jenkins
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FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num 2. Cert. Type 4. Author. By	
Manufacturer	Category	Model			Part	#: 066-1062-00	
Bendix/King	Transponder	KT 76A			Serial	#: 116770	
98-14-03 8/16/1998	TO PREVENT THE TRANSMISSION OF MISLEADING ENCODING ALTIMETER INFORMATION BETWEEN AFFECTED AIRCRAFT CAUSED	4-10-12 TACH 1767.6	NA by transponder SN. See AF log #1 for SN, 3-14-08, 642.5	Once	NA	1.University of Dubuque 2.AP 3.3015266 4. James S. Jenkins	
©ATP	BY THE, CONTOATP			©ATP	Signature:	Jamos. Julions	
Manufacturer	Category	Model			Part	₩:	
Am-Safe, Inc.	Safety Belts	Any Model			Serial	#:	
87-17-06 9/22/1987	TO ELIMINATE RESTRAINT SYSTEM CONNECTORS WITH THE INCORRECT DIMENSIONS, WHICH COULD ALLOW INADVERTENT OPENING,	4-10-12 TACH 1767.6	NA by part number	Once	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins	



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Content Revision: 4/6/2	012 File ID: N18	Aircraft Registration: N1866S						
FAA AD Number Effective Date	Description	Complied Date Time	Amendment Number Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num 2. Cert. Type 4. Author. By		
Manufacturer	Category	Model			Part	#:		
Pacific Scientific Co.	Co. Safety Belts SAFETY BELTS				Serial #:			
80-01-05 R1 7/10/1980	TO PREVENT FAILURE TO OPEN OF THE FLIGHT CREW AND ATTENDANTS' SEAT BELTS	267.6	NA by part number	Once	NA	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins		
©ATP	©ATP	\		©ATP	Signature:			
83-11-03 7/18/1983	TO PREVENT THE ENTRAPMENT OF OCCUPANTS CAUSED BY THE INABILITY TO RELEASE THE RESTRAINT SYSTEM ASSEMBLY	Таен	NA by part number	Once	NA (1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins		
©ATP	©ATP			©ATP	Signature.	James Jenlas		
87-20-05 10/26/1987	TO ELIMINATE LAP BELT ASSEMBLIES WITH BELT RETRACTOR SHAFTS WHICH DO NOT PROVIDE ADEQUATE STRENGTH	(O-/3	NA by part number and manufacture date	Once	NA /	1. University of Dubuque 2. AP 3. 3015266 4. James S. Jenkins		
©ATP	©ATP	*		©ATP	Signature:	amos Jenless		
94-21-06 11/25/1994	TO PREVENT THE INABILITY OF PASSENGERS OR CREW TO EGRESS FROM THEIR SEATS DURING AN EMERGENCY SITUATION, CONTD.		NA by part number	Once	NA	1. University of Dubuque 2. AP 3.3015266 4. James S. Jenkins		
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AIRWORTHINESS DIRECTIVE COMPLIANCE RECORD

Aircraft, engine, Propeller, or Appliance/Component Make

SOCATA Model TB-20

1866

AD Number & Rev. Date	RECURRING Subject AD'S	Date and hours at compliance	Method of Compliance	One - Time	Recurring	Next compliance due Date/Hours	Authorized Signature and Number
98-18-13	Main landing gear bearing	@ 4000	DUE @ 4000 hr ACTT		X	@4000	J.
10-24-98	LEFT side	Hr ACTT				Hr ACTT	20/25
05-12-06	Mag impulse coupling	8-14-2012	PCW by inspection		X	^	8/0
7-19-05		1851.0	SEE G&N AD LIST	N	X	2851.0	1.00
01-23-04	Rudder hinge	04-08-2014	PCW by insp lower hinge		X	O 1	
1-4-02		2247.0	SEE #1 AF LOG			4-2015	X '
03-04-03	Aileron gimbal joint	04-08-2014	PCW by insp		X	1 1	11/x
4-7-03		2247.0	SEE #1 AF LOG			2347.0	Ores or
11-26-04	Fuel injection lines	04-08-2014	PCW by insp	1	X	1	XX
1-25-12		2247.0	SEE #2 ENGLOG		4	2347.0	
2005-12-06	MAGNETO	8-19-14	INSP IMPULSE		X		July 1
		23461	coopling FOR WEAR		A	2846/	Tay
03-04-03	AilERON Gimbal	8-19-14	INSP GIRNOAL JOINT		X		The second
		2346.1	FOR WEAR			2446	243
11-26-04	FUEL MECTION	8-19-14	INSPFUEL LINES				10 12
		2346.1	+ clamps			2446	Ly of
03-04-03	AILERON GIMBAL	15/	FIMA GIMBAL JOINT		X		John o
-		13/1	FOR WEAR			2546,1	W. 12
11-26-04	FUEL INJECTION	3/4/00	INSO FUEL LINES		X		AND TES
		34	+ clamps			2546.1	101
01-23-04	Rudder hinge	7/14/2015	INSPECTED LOWER		X		Lry.
-		2545.4	HingE			7-2016	Parting
03-04-03	AILERON	7-14-15	INSP FOR WEAR +		X		12/38
	Gimbal	2545.4	SECURITY			2645	2/20
11-26-04	FUEL INJECTION	7-14-15	INSP FUEL LINES		X		10,76
		2545.4	+ clamps			2645	10 10 10 10 10 10 10 10 10 10 10 10 10 1
2005-12-	06 impulse	9-1-15	Jusp FOR WEAR +		X		Paul M Kou
	coupling	2621.4	SECURITY			3/2/	YOYR343B
	J		·			1	